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# Swing arm profilometer: analytical solutions of misalignment errors for testing axisymmetric optics 

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# Swing arm profilometer: analytical solutions of misalignment errors for testing axisymmetric optics 

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#### Abstract

The swing arm profilometer (SAP) has been playing a very important role in testing large aspheric optics. As one of most significant error sources that affects the test accuracy, misalignment error leads to low-order errors such as aspherical aberrations and coma apart from power. In order to analyze the effect of misalignment errors, the relation between alignment parameters and test results of axisymmetric optics is presented. Analytical solutions of SAP system errors from tested mirror misalignment, arm length $L$ deviation, tilt-angle $\theta$ deviation, air-table spin error, and air-table misalignment are derived, respectively; and misalignment tolerance is given to guide surface measurement. In addition, experiments on a 2-m diameter parabolic mirror are demonstrated to verify the model; according to the error budget, we achieve the SAP test for low-order errors except power with accuracy of $0.1 \mu \mathrm{~m}$ root-mean-square. © 2016 Society of Photo-Optical Instrumentation Engineers (SPIE) [DOI: 10.1117/1.OE.55.7.074108]


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## 1 Introduction

The swing arm profilometer (SAP) was first reported by Anderson et al. ${ }^{1,2}$ in the 1990s. The testing theory of the SAP is based on the fact that all measuring instruments have an accuracy that is proportional to their total range; to test an aspheric surface, the minimum testing range is the departure from its best-fit sphere. The way to achieve this is to move the sensor along the vertex sphere. Based on the geometry of a sphere generator, SAP was implemented with a sensor attached to the end of a radius rod pivoting to the center of the sphere. It is useful and highly efficient for large mirrors exceeding the range of a coordinate-measuring machine.

The SAP test has been playing a very important role in testing large aspheric optics. As the pioneer of the SAP test, the Arizona group has reported outstanding results for measuring large optics; ${ }^{1-7}$ Compared with the interferometer null test with the 43 low-order Zernike terms removed, the accuracy of SAP for testing $1.4-\mathrm{m}$ mirror is 5 nm in root-mean-square (RMS). Groups from London and Chengdu China have also done excellent work on SAP testing, achieving SAP testing for a $350-\mathrm{mm}$ diameter mirror within uncertainty of $0.16 \mu \mathrm{~m}$ in peak-valley (PV) and $0.02 \mu \mathrm{~m}$ in RMS $^{8-11}$ without low-order terms removed. The accuracy of other works from South Korea ${ }^{12}$ or Changsha China ${ }^{13}$ was reported to be microns in PV. Changchun Institute of Optics, Fine Mechanics, and Physics has achieved success in measuring an aspheric mirror up to 2 m in diameter by SAP with an accuracy of $\sim 2 \mu \mathrm{~m}$ in PV without low-order terms removed except power. ${ }^{14}$

For improving the test accuracy, the important guarantee is supposed to be alignment and calibration. ${ }^{5,6,15}$ An ideal

SAP test is to align SAP in a position related to the mirror under test and assure the trajectory of the sensor probe on best-fit sphere (BFS) of optic, while in fact, SAP cannot be aligned in a perfect way due to alignment error. It turns out that alignment error is an important error source that leads to low-order errors such as aspherical aberrations, astigmatism, and coma apart from power. SAP is used to measure optic surface during the grinding process, ${ }^{7,11,15-17}$ when surface accuracy comes to 1 to $2 \mu \mathrm{~m}$ in PV ; the effect of misalignment except for power comes to show up and cannot be ignored.

In this paper, we present how misalignment of SAP components impact on tests of axisymmetric optical. The analysis has been applied for a 2-m diameter parabolic surface. Model setup and analysis are presented in Sec. 2 as follows. Basic principle of SAP is first described in Sec. 2.1. An analytical model is built as relation between axisymmetric surface and alignment parameters of SAP given in Sec. 2.2. Then by performing partial derivative and series expansion, the solutions of errors that misalignment elements produced on surface test result are derived in Sec. 2.3. Based on the solutions, misalignment tolerance for the SAP test of 2-m mirror is proposed in Sec. 2.4. Experiment and verification are reported in Sec. 3. The verification of the analytical model is presented with experiment on a misalignment parameter in Sec. 3.1. The verification of SAP test compared with interferometer test is presented in Sec. 3.2. The conclusion is given in Sec. 4.

The paper is aimed to present the preparatory work for decoupling misalignment errors from test results in loworder terms. Further analysis of misalignment errors for testing off-axis aspheric optics are in process.


Fig. 1 Basic principle of SAP test: (a) geometry of SAP and (b) scan pattern.

## 2 Model and Analysis

### 2.1 Basic Principle of Swing Arm Profilometer

The basic geometry of the SAP ${ }^{1-3}$ is shown in Fig. 1(a). A probe is mounted at the end of an arm that is fixed on a high accuracy air-table whose rotation axis is tilted and goes through the center of BFS. By rotating the mirror, twodimensional profile can be obtained with multiple scans as shown in Fig. 1(b).The tilt-angle $\theta$ of air-table is given by
$\theta=\sin ^{-1}\left(\frac{L}{R_{\mathrm{bfs}}}\right)$,
where $L$ is the distance between probe tip and rotation axis of air-table and $R_{\text {bfs }}$ is the radius of curvature of BFS.

The surface result is given by $\Delta S=S_{t}-S_{0}$, where $S_{t}$ is the probe reading and $S_{0}$ is the ideal aspherical departure from the BFS.

### 2.2 Model Setup

An SAP has three components: mirror (on turntable), probe, and air-table, which carries the probe. As shown in Fig. 1(a),

Table 1 Alignment requirement of SAP.

| Items | Alignment goal | Error source |
| :--- | :--- | :--- |
| Mirror | Center of mirror coincide with <br> the rotation axis of turntable | Decentration |
| Air-table | Rotation axis of air-table turntable <br> goes through $C_{\text {bfs }}$ | Tilt, decentration |
| Probe | Detect direction points to $C_{\text {bfs }}$ | Tilt |
|  | Coincides with the geometric center <br> of mirror at null position | Decentration |
| ${ }^{\text {a }}$ Null position: probe point to the center of mirror as shown in Fig. 1(a). |  |  |

the rotation axis of the turntable coincides with the center of BFS; the air-table is tilted and its rotation axis goes through the center of BFS; the probe is aligned at the end of the arm and its detect direction points to $C_{\mathrm{bfs}}$ so that the probe would measure the aspheric departure from BFS. The error sources of the components are listed in Table 1. In this section, the test result of SAP for axisymmetric asphere will be described in a geometry model, and the analytic solution will be explained.

### 2.2.1 Geometry of asphere

The geometry of points $P$ on axisymmetric optic ${ }^{1}$ is shown in Fig. 2, and given by

$$
\left\{\begin{array}{l}
r^{2}-2 R z+(k+1) z^{2}=0(\text { aspheric-equation })  \tag{2}\\
S \cdot \cos (\xi)+z=R_{\mathrm{bfs}} \\
S \cdot \sin (\xi)=r
\end{array}\right.
$$



Fig. 2 Distance between surface of axisymmetric asphere and center of BFS.
where $r$ is the distance of point $P$ to optic axis, $R$ is the radius of curvature at vertex, $k$ is the conic constant, and $\xi$ is the angle of elevation of point $P$.

Based on formulas in Eq. (2), we get the aspheric departure from BFS associated with $\xi$ by

$$
\begin{align*}
S_{0} & =F(\xi)=R_{\mathrm{bfs}}-S \\
& =R_{\mathrm{bfs}}-\frac{2 R R_{\mathrm{bfs}}-(1+k) R_{\mathrm{bfs}}^{2}}{K_{1} \cos (\xi)+\sqrt{R^{2}-K_{2} \sin ^{2}(\xi)}}, \tag{3}
\end{align*}
$$

where $K_{1}=R-R_{\mathrm{bfs}}(k+1), K_{2}=R_{\mathrm{bfs}}^{2}(k+1)-2 R R_{\mathrm{bfs}}+R^{2}$.

### 2.2.2 Analytical model of Swing Arm Profilometer

The coordinate systems and parameters in the SAP model are shown in Fig. 3. Coordinate system $O X Y Z$ and $O_{2} X_{2} Y_{2} Z_{2}$ are both in right-hand coordinate.

The parameters of the SAP system are described below:
$O X Y Z$ represents SAP test coordinate system;
$O_{2} X_{2} Y_{2} Z_{2}$ represents air-table coordinate system;
$P^{\prime}$ is the point on spherical track of SAP test;
$P^{\prime \prime}$ is projection on $X Y$-plane of point $P^{\prime}$;
$\beta$ is air-table spin angle;
$\alpha$ is rotate angle of turntable;
$\rho$ is polar radius of test point $P^{\prime}$;
$\eta^{\prime}$ is polar angle of point $P^{\prime \prime}$.
The coordinate of points $P^{\prime}$ in the $O X Y Z$ system is expressed by
$x=R_{\mathrm{bfs}} \cdot \sin (\xi) \cdot \cos \left(\eta^{\prime}\right)$,
$y=R_{\mathrm{bfs}} \cdot \sin (\xi) \cdot \sin \left(\eta^{\prime}\right)$,
$z=R_{\mathrm{bfs}}-R_{\mathrm{bfs}} \cdot \cos (\xi)$.


Fig. 3 Parameters of SAP system.


Fig. 4 Flowchart of analytical model.

According to knowledge of triangular geometry, we get
$\rho=2 L \cdot \sin \left(\frac{\beta}{2}\right) \cdot \cos \left(\frac{\pi-\xi}{2}\right)$,
$\xi=G\left(L, \beta, R_{\mathrm{bfs}}\right)=2 \sin ^{-1}\left(\frac{L \cdot \sin (\beta / 2)}{R_{\mathrm{bfs}}}\right) ; \quad \beta \in(0,2 \pi)$,
$\eta=\tan ^{-1}\left(\frac{L \cdot \sin (\beta)}{L \cdot[1-\cos (\beta)] \cdot \cos (\theta)}\right)$.
As the mirror turntable rotated angle $\alpha$, we get $\eta^{\prime}=\eta+\alpha$. In the SAP test system, the relationship between test results and parameters of SAP could be built through Eqs. (3), (6), and (7). Based on this, misalignment problems are discussed in detail in Sec. 2.3, and explicit solutions are derived by performing partial derivative and series expansion, as the flowchart of process shown in Fig. 4.

### 2.3 Misalignment Analysis

### 2.3.1 Mirror misalignment error

In a spherical coordinate system, coordinate of points $P$ on a mirror as shown in Fig. 2 is given by
$x=r \cdot \cos (\psi) ; \quad y=r \cdot \sin (\psi)$,
where $r$ is the polar radius and $\psi$ is the polar angle in $X Y$-plane. According to the knowledge of perfect differential, we get

$$
\begin{equation*}
\mathrm{d} r=\cos (\psi) \mathrm{d} x+\sin (\psi) \mathrm{d} y \tag{9}
\end{equation*}
$$

$\mathrm{d} \psi=-\frac{\sin (\psi)}{r} \mathrm{~d} x+\frac{\cos (\psi)}{r} \mathrm{~d} y$.
The axisymmetric optical surface is represented by

$$
\begin{equation*}
z=\frac{r^{2}}{R\left[1+\sqrt{1-(k+1) \frac{r^{2}}{R^{2}}}\right]} \tag{10}
\end{equation*}
$$

Meanwhile, by performing series expansion, the departure toward $C_{\mathrm{bfs}}$ is represented by

$$
\begin{align*}
S_{0} & =R_{\mathrm{bfs}}-\sqrt{x^{2}+y^{2}+\left(z-R_{\mathrm{bfs}}\right)^{2}} \\
& =c_{1} r^{2}+c_{2} r^{4}+o\left(r^{4}\right), \tag{11}
\end{align*}
$$

where
$c_{1}=-\frac{R-R_{\mathrm{bfs}}}{2 R R_{\mathrm{bfs}}} ;$
$c_{2}=-\frac{R-R_{\mathrm{bfs}}}{R_{\mathrm{bfs}}} \cdot \frac{1+k}{8 R^{3}}+\left[\frac{\left(R-R_{\mathrm{bs}}\right)^{2}}{R_{\mathrm{bfs}}^{3}}+\frac{k}{R_{\mathrm{bfs}}}\right] \frac{1}{8 R^{2}}$,
$o\left(r^{4}\right)$ is fourth minimum order of $r$.
The effect of mirror decentration is obtained according to
$\frac{\partial S_{0}}{\partial x}=\frac{\partial S_{0}}{\partial r} \cdot \frac{\partial r}{\partial x}=\left(2 c_{1} r+4 c_{2} r^{3}\right) \cos (\psi)+o\left(r^{3}\right)$,
$\frac{\partial S_{0}}{\partial y}=\frac{\partial S_{0}}{\partial r} \cdot \frac{\partial r}{\partial y}=\left(2 c_{1} r+4 c_{2} r^{3}\right) \sin (\psi)+o\left(r^{3}\right)$.
It is known from Eq. (12) that deviation of the axisymmetric mirror in $X$-axis and $Y$-axis result in 0 and 90 deg coma, respectively.

For a real parabolic mirror, whose $k=-1, \quad D=$ $2000 \mathrm{~mm}, R=6000 \mathrm{~mm}, R_{\mathrm{bfs}}=6100 \mathrm{~mm}$, its deviation in $X$-axis direction is 1 mm . Ignore tilt, and normalize $r$ by $r_{\text {max }}=D / 2$, we get the system error allocated to coma terms expressed as $\left[r\left(-2+3 r^{2}\right) \cdot \cos \psi\right]$ by
$Z_{\text {coma }}=\frac{4 c_{2}}{3} r_{\text {max }}^{3} \delta x=7.587 \times 10^{-4} \mathrm{~mm}$.
It means that per 1 mm decentration of $2-\mathrm{m}$ diameter polibolid along $X$-axis produces coma in $1.5 \mu \mathrm{~m} \mathrm{PV}$ and $0.27 \mu \mathrm{~m}$ RMS.

### 2.3.2 Arm L deviation

In the SAP system, by performing Taylor expansion, $S(\xi)$ in Eq. (3) can be represented by
$S_{0}(\xi)=a_{1} \xi^{2}+a_{2} \xi^{4}+a_{3} \xi^{6}+o\left(\xi^{6}\right)$,
where
$a_{1}=\frac{\left(K_{1}+K_{2} / R\right) R_{\mathrm{bfs}}}{2\left(K_{1}+R\right)} ;$
$a_{2}=\left[-\frac{\left(-K_{1}-K_{2} / R\right)^{2}}{4\left(K_{1}+R\right)^{2}}+\frac{K_{1}-3 K_{2}^{2} / R^{3}+4 K_{2} / R}{24\left(K_{1}+R\right)}\right] \cdot R_{\mathrm{bfs}}$.
On the other hand, by eliminating $\xi$ from Eqs. (5) and (6), the relationship between $\beta$ and $\rho$ is obtained as

$$
\begin{equation*}
\left[L \cdot \sin \left(\frac{\beta}{2}\right)\right]^{2}=\frac{R_{\mathrm{bfs}}}{2}\left(R_{\mathrm{bfs}}-\sqrt{R_{\mathrm{bfs}}^{2}-\rho^{2}}\right) . \tag{14}
\end{equation*}
$$

With substitution for $\beta$ from Eq. (14), the differential relationship between $\xi$ and $L$ according to Eq. (6) is given by

$$
\begin{equation*}
\frac{\partial \xi}{\partial L}=\frac{2 \sin (\beta / 2)}{\sqrt{R_{\mathrm{bfs}}^{2}-L^{2} \sin ^{2}(\beta / 2)}}=\frac{2}{L\left(R_{\mathrm{bfs}}+\sqrt{R_{\mathrm{bfs}}^{2}-\rho^{2}}\right)} . \tag{15}
\end{equation*}
$$

With $\xi=\sin ^{-1}\left(\rho / R_{\text {bfs }}\right)$ and by performing Taylor expansion, we get partial derivative about $\rho$ according to

$$
\begin{align*}
\frac{\partial S_{0}}{\partial L}= & \frac{\partial F}{\partial \xi} \cdot \frac{\partial \xi}{\partial L}=\left[2 a_{1} \xi+4 a_{2} \xi^{3}+6 a_{3} \xi^{5}+o\left(\xi^{5}\right)\right] \\
& \cdot \frac{2}{L\left(R_{\mathrm{bfs}}+\sqrt{R_{\mathrm{bfs}}^{2}-\rho^{2}}\right)}, \\
\frac{\partial S_{0}}{\partial L}= & b_{1} \rho^{2}+b_{2} \rho^{4}+o\left(\rho^{4}\right), \tag{16}
\end{align*}
$$

where
$b_{1}=\frac{2 a_{1}}{L \cdot R_{\mathrm{bfs}}^{2}} ; \quad b_{2}=\frac{5 a_{1}+24 a_{2}}{6 L \cdot R_{\mathrm{bfs}}^{4}}$.
For axisymmetric mirror, from Eq. (16), it is known that deviation of $L$ results in defocus and spherical errors. For a $2-\mathrm{m}$ diameter parabolic mirror, $L=500 \mathrm{~mm}, \delta L=1 \mathrm{~mm}$; normalizing $\rho$ by half diameter of mirror, which is 1000 mm , we get

$$
\begin{align*}
\Delta S_{L}= & \frac{\partial S_{0}}{\partial L} \cdot \delta L=-0.786 \times 10^{-3}\left(1-6 \rho^{2}+6 \rho^{4}\right) \\
& -5.93 \times 10^{-2} \rho^{2}+0.786 \times 10^{-3} ; \quad \text { here, } \rho \in[0,1] \tag{17}
\end{align*}
$$

With respect to the normalized Zernike terms of $\mathrm{SA}_{3}$ (primary spherical aberration), expressed as $\left(1-6 \rho^{2}+6 \rho^{4}\right)$, the value of $\mathrm{SA}_{3}$ in $\Delta S_{L}$ is $-0.786 \mu \mathrm{~m}$.

It means that $L$ deviation per 1 mm results in $\mathrm{SA}_{3}$ error of $1.179 \mu \mathrm{~m}$ in PV, which is 1.5 times the value of $\mathrm{SA}_{3}$. Here, a simulation is presented. We first tested the $2-\mathrm{m}$ diameter parabolic mirror by SAP as the result shown in Fig. 5(a). Then, $1-\mathrm{mm}$ deviation is joined into $L$ during data processing and the test result is shown in Fig. 5(b). Difference between two maps is $1.184 \mu \mathrm{~m}$ in PV and $0.267 \mu \mathrm{~m}$ in RMS as shown in Fig. 5(c). The simulated result of $1.184 \mu \mathrm{~m}$ differs $0.4 \%$ from calculated result of $1.179 \mu \mathrm{~m}$ which is negligible and verified each other.

### 2.3.3 Tilt-angle $\theta$ deviation

According to Eq. (1), we get
$\Delta S_{\theta}=\frac{\partial S_{0}}{\partial \theta} \cdot \delta \theta=\frac{\partial S_{0}}{\partial L} \cdot R_{\mathrm{bfs}} \cdot \cos (\theta) \cdot \delta \theta$.
It means that the influence from tilt-angle deviation has the same form with the one from Arm $L$ errors with a factor of $R_{\mathrm{bfs}} \cos \theta$. In the example above, per 1-mm deviation of $L$ equals to about 0.01 deg deviation of $\theta$.

### 2.3.4 Rotation error $\delta \beta$ of air-table

As presented in Table 1, the probe should be aligned to coincide with the center of mirror at null position shown


Fig. 5 Simulation of $L$ deviation: (a) initial SAP test result $\mathrm{PV}=3.743 \mu \mathrm{~m}, \mathrm{RMS}=0.45 \mu \mathrm{~m}$; (b) SAP test result with $L$ deviated $1 \mathrm{~mm} \operatorname{PV}=3.724 \mu \mathrm{~m}$, $\mathrm{RMS}=0.605 \mu \mathrm{~m}$; (c) difference $\mathrm{PV}=1.184 \mu \mathrm{~m}$, RMS $=0.267 \mu \mathrm{~m}$.
in Fig. 1(a). The length of the arm is adjusted to assure the alignment in $X$-axis direction, and the spin angle $\beta$ is adjusted to assure the alignment in $Y$-axis direction. So the decentration in $Y$-axis direction of probe means a constant delayed or advanced error of angle $\beta$. For axisymmetric optics, the terms of air-table errors in different angle $\alpha$ are the same. For easy description, we discuss the case in single scan, where $\alpha=0$ deg. According to Eq. (6), we get the partial derivative of $\beta$ by

$$
\begin{equation*}
\frac{\partial \xi}{\partial \beta}=\frac{L \cos (\beta / 2)}{\sqrt{R_{\mathrm{bfs}}^{2}-L^{2} \sin ^{2}(\beta / 2)}} . \tag{19}
\end{equation*}
$$

So according to Eq. (3), we obtain

$$
\begin{align*}
\frac{\partial S_{0}}{\partial \beta}= & \frac{\partial F}{\partial \xi} \cdot \frac{\partial \xi}{\partial \beta}=-\frac{2 R R_{\mathrm{bfs}}-(1+k) R_{\mathrm{bfs}}^{2}}{\left[K_{1} \cos (\xi)+\sqrt{R^{2}-K_{2} \sin ^{2}(\xi)}\right]^{2}} \\
& \cdot\left[K_{1} \sin (\xi)+\frac{K_{2} \sin (2 \xi)}{2 \sqrt{R^{2}-K_{2} \sin ^{2}(\xi)}}\right] \\
& \cdot \frac{L \cos (\beta / 2)}{\sqrt{R_{\mathrm{bfs}}^{2}-L^{2} \sin ^{2}(\beta / 2)}} . \tag{20}
\end{align*}
$$

For the parabolic mirror, as alignment accuracy of probe along $Y$-axis direction $\delta y=1 \mathrm{~mm}$, which means $\delta \beta \approx \delta y / L=2$ mrad, the errors on single scan line with tilt removed are shown in Fig. 6(a). By rotating single scan around axis of mirror turntable, errors in the whole surface are given. Due to noncoincidence at crossings of neighboring scan lines, fitting effect on the surface is in the form of random noise as shown in Fig. 6(b) with $0.2 \mu \mathrm{~m}$ in PV and $0.01 \mu \mathrm{~m}$ in RMS.

### 2.3.5 Air-table misalignment errors

Approximation of swing arm profilometer test. In the coordinate system $\mathrm{O}_{2} X_{2} Y_{2} Z_{2}$ as shown in Fig. 3, the misalignment errors of air-table mainly include displacements along $X_{2}$-axis and $Y_{2}$-axis direction, and rotary deviations about $X_{2}$-axis and $Y_{2}$-axis.

As a matter of fact, air-table displacement along $X_{2}$-axis is equivalent to the deviation of parameter $L$, and air-table rotary deviation about $Y_{2}$-axis is equivalent to the deviation of parameter $\theta$ of SAP system, which resulted in defocus and aspherical aberrations. Air-table displacement along $Y_{2}$-axis and rotary deviation about $X_{2}$-axis deflect the rotary axis of


Fig. 6 Errors result from $\beta$ deviation: (a) error result in single scan data with tilt removed and (b) error result in whole surface.
air-table from rotary axis of mirror turntable, which is mainly discussed later.

In the SAP test model, substituting for $\xi=\sin ^{-1}\left(\rho / R_{\mathrm{bfs}}\right)$, the aspherical departure in Eq. (3) becomes
$\Delta S_{0}=R_{\mathrm{bfs}}-\frac{2 R R_{\mathrm{bfs}}-(1+k) R_{\mathrm{bfs}}^{2}}{K_{1} \sqrt{1-\rho^{2} / R_{\mathrm{bfs}}^{2}}+\sqrt{R^{2}-K_{2} \rho^{2} / R_{\mathrm{bfs}}^{2}}}$.

The approximate aspherical departure is used in some cases $^{8,10}$ as given by
$S_{\mathrm{appr}}=R_{\mathrm{bfs}}-\sqrt{\rho^{2}+\left[z(\rho)-R_{\mathrm{bfs}}\right]^{2}}$,
where

$$
z=\frac{\rho^{2}}{R\left(1+\sqrt{1-(k+1) \rho^{2} / R^{2}}\right)} .
$$

Approximation error is given in the example of 2-m diameter parabolic mirror. The exact aspherical departure is $142.385 \mu \mathrm{~m}$ in PV as shown in Fig. 7(a) and the approximate one is $142.443 \mu \mathrm{~m}$ in PV as shown in Fig. 7(a). The difference between exact solution and approximate one is about 49 nm in PV and $0.017 \mu \mathrm{~m}$ in RMS with defocus removed as shown in Fig. 7(c). Since the approximate error is acceptable, the simplified expression of aspheric departure in Eq. (22) will be used in the work below.

Air-table rotatory errors about $\boldsymbol{X}_{2}$-axis. In the SAP model shown in Fig. 3, after air-table rotated a tiny angle $u$ about $X_{2}$, the coordinate $(x, y, z)$ of points $P$ on the single scan line $(\alpha=0)$ turns out to be ( $x^{\prime}, y^{\prime}, z^{\prime}$ ) as follows:
$\left(\begin{array}{l}x^{\prime} \\ y^{\prime} \\ z^{\prime}\end{array}\right)=r_{y}^{\prime} \cdot r_{x(u)} \cdot r_{y} \cdot\left(\begin{array}{l}x \\ y \\ z\end{array}\right)$,
where

$$
\begin{aligned}
r_{y} & =\left(\begin{array}{ccc}
\cos (\theta) & 0 & \sin (\theta) \\
0 & 1 & 0 \\
-\sin (\theta) & 0 & \cos (\theta)
\end{array}\right), \\
r_{y}^{\prime} & =\left(\begin{array}{ccc}
\cos (-\theta) & 0 & \sin (-\theta) \\
0 & 1 & 0 \\
-\sin (-\theta) & 0 & \cos (-\theta)
\end{array}\right), \\
r_{x(u)} & =\left(\begin{array}{ccc}
1 & 0 & 0 \\
0 & \cos (u) & -\sin (u) \\
0 & \sin (u) & \cos (u)
\end{array}\right)
\end{aligned}
$$

As $u$ is a minuteness, we obtain

$$
\left(\begin{array}{l}
x^{\prime}  \tag{24}\\
y^{\prime} \\
z^{\prime}
\end{array}\right) \cong\left(\begin{array}{l}
x-u \cdot y \cdot \sin \theta \\
y-z \cdot \cos \theta \cdot \sin u+x \cdot \sin \theta \cdot \sin u \\
z+u \cdot y \cdot \cos \theta
\end{array}\right)
$$

Meanwhile, center of SAP sphere changed from ( $0,0, R_{\mathrm{bfs}}$ ) to $\left(0, \delta y, R_{\mathrm{bfs}}\right)$, where $\delta y=-u R_{\mathrm{bfs}} \cos (\theta)$, then the aspheric departure tested by SAP in a single scan line becomes

$$
\begin{align*}
S^{\prime} & =f(x, y, u) \\
& =R_{\mathrm{bfs}}-\sqrt{x^{\prime 2}+\left(y^{\prime}-\delta y\right)^{2}+\left[z\left(x^{\prime}, y^{\prime}\right)-R_{\mathrm{bfs}}\right]^{2}} \\
& \approx s^{\prime}(u=0)+\frac{\partial S^{\prime}}{\partial u} \cdot u . \tag{25}
\end{align*}
$$

According to Eq. (22) and Eq. (9), where $\rho^{\prime}$ here means the polar radius $r$ there, by performing partial derivative, we obtain

$$
\begin{align*}
\left.\frac{\partial z\left(x^{\prime}, y^{\prime}\right)^{\prime}}{\partial u}\right|_{u=0} & =\left.\frac{\partial z}{\partial \rho^{\prime}}\left(\frac{\partial \rho^{\prime}}{\partial x^{\prime}} \cdot \frac{\partial x^{\prime}}{\partial u}+\frac{\partial \rho^{\prime}}{\partial y^{\prime}} \cdot \frac{\partial y^{\prime}}{\partial u}\right)\right|_{u=0} \\
& =\frac{-\rho z \cos (\theta)}{\sqrt{R^{2}-(1+k) \rho^{2}}} \tag{26}
\end{align*}
$$

Here,
$z=R_{\mathrm{bfs}}-\sqrt{R_{\mathrm{bfs}}^{2}-\rho^{2}}$.
The analytical solution of the test error from air-table rotatory deviation about $X_{2}$-axis is given by
$S_{u}=\left.\frac{\partial S^{\prime}}{\partial u}\right|_{u=0} \cdot u=\frac{x y \sin \theta-y\left(x \sin \theta-\sqrt{R_{\mathrm{bfs}}^{2}-\rho^{2}} \cos \theta+R_{\mathrm{bfs}} \cos \theta\right)+\left[z(\rho)-R_{\mathrm{bfs}}\right] \frac{\rho z \cos \theta}{\sqrt{R^{2}-(1+k) \rho^{2}}}}{\sqrt{\rho^{2}+\left[z(\rho)-R_{\mathrm{bfs}}\right]^{2}}} u$.

The consequent error of $u$ being 0.01 deg is calculated, as error in a single scan with tilt removed shown in Fig. 8(a). By rotating the single scan around axis of mirror turntable, the effect on the whole surface is mainly in a $\mathrm{SA}_{3}$ form in $0.43 \mu \mathrm{~m}$ PV and $0.012 \mu \mathrm{~m}$ RMS as shown in Fig. 8(b).

Air-table displacement along $\boldsymbol{Y}_{2}$-axis. After air-table displaced $\mu$ along $Y_{2}$-axis, the data transformation matrix is given by
$\left(\begin{array}{l}x^{\prime} \\ y^{\prime} \\ z^{\prime}\end{array}\right)=\left(\begin{array}{l}x \\ y+\mu \\ z\end{array}\right)$.
The tested aspheric departure in Eq. (22) of a single scan line becomes

$$
\begin{align*}
S^{\prime} & =f(x, y, u) \\
& =R_{\mathrm{bfs}}-\sqrt{x^{\prime 2}+\left(y^{\prime}-\delta y\right)^{2}+\left[z\left(x^{\prime}, y^{\prime}\right)-R_{\mathrm{bfs}}\right]^{2}} \tag{29}
\end{align*}
$$


(a)

Here, $\delta y=\mu$. In a similar way explained in part B , we get

$$
\begin{align*}
& \left.\frac{\partial z\left(x^{\prime}, y^{\prime}\right)^{\prime}}{\partial u}\right|_{u=0}=\frac{y}{\sqrt{R^{2}-(1+k) \rho^{2}}},  \tag{30}\\
& S_{\mu}=\left.\frac{\partial S^{\prime}}{\partial \mu}\right|_{\mu=0} \cdot \mu=-\frac{z(\rho)-R_{\mathrm{bfs}}}{\sqrt{\rho^{2}+\left[z(\rho)-R_{\mathrm{bfs}}\right]^{2}}} \cdot \frac{y}{\sqrt{R^{2}-(1+k) \rho^{2}}} \cdot \mu . \tag{31}
\end{align*}
$$

The consequent errors of $\mu$ being 0.01 deg are shown in Fig. 9. Error of single scan data with tilt off is shown in Fig. 9(a); due to fitting effect, influence on the whole surface is in the form of random noise with $0.18 \mu \mathrm{~m}$ in PV and $0.01 \mu \mathrm{~m}$ in RMS as shown in Fig. 9(b).

### 2.4 Misalignment Tolerance

Related to the probe range, probe disvertical from surface will result in a small displacement in $X / Y$-axis and a small quantity of the second order in probe readout,


Fig. 7 Comparison of the approximate departure of parabolid from exact one: (a) the exact departure; (b) the approximate departure; and (c) difference.


Fig. 8 Influence in test result of air-table with rotary deviation about $X_{2}$-axis: (a) errors on single scan data with tilt off and (b) errors on the whole surface.


Fig. 9 Influence in test result of air-table displaced along $Y_{2}$-axis: (a) errors on single scan data with tilt off and (b) errors on the whole surface.

Table 2 Alignment budget of SAP for 2-m diameter parabolic.

|  | $>5 \mu \mathrm{mPV}, 1 \mu \mathrm{~m} \mathrm{RMS}$ |  | 0.5 to $5 \mu \mathrm{mPV} ;<1 \mu \mathrm{~m}$ RMS |  |
| :---: | :---: | :---: | :---: | :---: |
| Accuracy of surface | Misalignment tolerance | Error in RMS/ $\mu \mathrm{m}$ | Misalignment tolerance | Error in RMS/ $\mu \mathrm{m}$ |
| Mirror alignment | 1 mm | 0.27 (coma) | 0.2 mm | 0.05 (coma) |
| Length of arm L | 1 mm | 0.27 ( $\left.\mathrm{SA}_{3}\right)$ | 0.1 mm | 0.03 ( $\mathrm{SA}_{3}$ ) |
| Tilt-angle $\theta$ | 0.01 deg | $0.27\left(\mathrm{SA}_{3}\right)$ | 0.001 deg | $0.03\left(\mathrm{SA}_{3}\right)$ |
| Probe alignment Along arm | 0.5 mm | $0.14\left(\mathrm{SA}_{3}\right)$ | 0.1 mm | $0.03\left(\mathrm{SA}_{3}\right)$ |
| Perpendicular direction of $\mathrm{arm}^{\text {a }}$ | 1 mm | 0.01 (random) | 0.05 mm | 0.001 (random) |
| Air-table posture Rotate about $X_{2}{ }^{\text {b }}$ | 0.01 deg | $0.12\left(\mathrm{SA}_{3}\right)$ | 0.001 deg | $0.012\left(\mathrm{SA}_{3}\right)$ |
| Displaced along $Y_{2}$ | 1 mm | 0.01 (random) | 0.001 deg | 0.001 (random) |
| Total |  | 0.84 |  | 0.11 |

[^0]

Fig. 10 SAP setup for 2-m diameter paraboloid.
which can be ignored here. From the analyses of the SAP model, the errors from misalignments of SAP for testing axisymmetric surface are in the form of $\mathrm{SA}_{3}$ and random noise.

It is an expensive and hard work to achieve strict alignment budget. During the rough grinding process, loose alignment is acceptable and more efficient; whereas strict budget is necessary for a higher accuracy surface test. In two phases of different accuracy of surface, misalignment tolerances of SAP for 2-m diameter parabolic are allocated as shown in Table 2.

## 3 Experiments on 2-m Diameter Parabolic Mirror

In these parts, a particular SAP ${ }^{14,16}$ is used to test the $2-\mathrm{m}$ diameter parabolic mirror as an experiment. As SAP setup shown in Fig. 10, the air-table is fixed on the spindle of a grinding/polishing machine, and 2-m diameter mirror is placed on the turntable of the machine. The spindle is used to generate the tilt-angle $\theta$ of SAP, and $X Y Z$ slide axes are used to adjust the position of SAP.

### 3.1 Experiment Verification of Misalignment Analysis

Due to space limitation, here, we just perform an experiment of air-table displacement along $X$-axis to verify the analytical model. As explained in Sec. 2.3.5.1, air-table displacement along $X$-axis is equivalent to the displacement of arm length $L$. In this experiment, air-table is displaced 2 mm along + $X$-axis to test the misalignment sensitivity of SAP. The influence of displacement in test results is shown in Fig. 11. The initial test result of SAP is shown in Fig. 11(a); the test result after displacement is shown in Fig. 11(b); and their difference is $\mathrm{SA}_{3}$ errors in $2.325 \mu \mathrm{~m} \mathrm{PV}$ as shown in Fig. 11(c). As described in Sec. 2.3.2, per $1 \mathrm{~mm} L$ displacement produces $1.179 \mu \mathrm{~m} \mathrm{PV} \mathrm{SA}_{3}$ errors. So the percentage error of experiment results from the calculated solution is $1.4 \%$, which is acceptable. The analytical model is checked as the result coinciding with analytical solution.


Fig. 11 Influence in test result of air-table displaced along $X$-axis: (a) before displacement, $\mathrm{PV}=2.247 \mu \mathrm{~m}, \mathrm{RMS}=0.352 \mu \mathrm{~m}$; (b) after displacement, $\mathrm{PV}=4.324 \mu \mathrm{~m}, \mathrm{RMS}=0.774 \mu \mathrm{~m}$; and (c) difference, $\mathrm{PV}=2.325 \mu \mathrm{~m}$, $\mathrm{RMS}=0.364 \mu \mathrm{~m}$.


Fig. 12 SAP test result for 2-m diameter mirror, $\mathrm{PV}=1.495 \mu \mathrm{~m}$, RMS $=0.166 \mu \mathrm{~m}$.


Fig. 13 Interferometer test result for 2-m diameter mirror, $\mathrm{PV}=$ $1.618 \mu \mathrm{~m}, \mathrm{RMS}=0.226 \mu \mathrm{~m}$.

### 3.2 Verified with Interferometer Test

The interferometer is mounted above the mirror in the test tower. To verify the test accuracy of SAP, the mirror is polished and is measured by the interferometer in the same posture. SAP setup is calibrated according to the tolerance given in Table 2, and the test result is $1.495 \mu \mathrm{~m}$ in PV, $0.166 \mu \mathrm{~m}$ in RMS as shown in Fig. 12. The interferometer test result is $1.618 \mu \mathrm{~m}$ in PV and $0.226 \mu \mathrm{~m}$ in RMS as shown in Fig. 13. A direct subtraction between the SAP test result and the interferometer test result without any loworder terms removed expect power shows the difference of $\sim 0.857 \mu \mathrm{~m}$ in PV and $0.167 \mu \mathrm{~m}$ in RMS as shown in Fig. 14.

As analyzed in Sec. 2, misalignments of SAP for testing axisymmetric surfaces mainly produce coma and $\mathrm{SA}_{3}$ errors. Here, as shown in Fig. 15, the coma and $\mathrm{SA}_{3}$ errors in the deviation between SAP and the interferometer test result are $0.444 \mu \mathrm{~m} \mathrm{PV}$ and $0.108 \mu \mathrm{~m}$ RMS within the limits of error budget $0.11 \mu \mathrm{~m}$ RMS given in Table 2. The experiment


Fig. 14 The difference between SAP and Interferometer test result, $\mathrm{PV}=0.857 \mu \mathrm{~m}, \mathrm{RMS}=0.167 \mu \mathrm{~m}$.


Fig. 15 Coma and $\mathrm{SA}_{3}$ terms in the difference between SAP and interferometer test, $\mathrm{PV}=0.444 \mu \mathrm{~m}, \mathrm{RMS}=0.108 \mu \mathrm{~m}$.
result verified the error budget well. It means that according to the error budget, we achieve the SAP test for low-order errors with accuracy of $0.1 \mu \mathrm{~m}$ RMS. Due to the different sampling density, some high-frequency details in Figs. 13 and 14 do not match well. Other unexpected middle and high-frequency errors are caused by errors such as sensor probe read error and air-table and mirror turntable vibration errors.

## 4 Conclusion

The misalignment errors of SAP have been discussed in detail and analytical solutions are derived. The sensitivity ratios of misalignment for testing a $2-\mathrm{m}$ diameter paraboloid are calculated: per $L$ deviation of 1 mm brings $\mathrm{SA}_{3}$ in $1.2 \mu \mathrm{~m}$ PV , and 0.01 deg tilt-error of air-table around $X_{2}$ brings mainly $\mathrm{SA}_{3}$ error in $0.4 \mu \mathrm{~m} \mathrm{PV}$; while $1-\mathrm{mm}$ displacement of air-table along $Y_{2}$ and 0.1 deg delay of air-table rotary angle both lead to random error of $0.2 \mu \mathrm{~m}$ in PV. Alignment tolerance is allocated to assure the test accuracy
and verified experiments for the model are carried out. According to the error budget, we achieve the SAP test for low-order errors with accuracy of $0.1 \mu \mathrm{~m}$ RMS. Compared with the interferometer test result, the accuracy of the SAP test on a $2-\mathrm{m}$ diameter mirror is $0.9 \mu \mathrm{~m}$ in PV and $0.2 \mu \mathrm{~m}$ in RMS (only power term removed). Further work will focus on calibration of the middle and high-frequency errors to improve the test accuracy.

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[^0]:    ${ }^{\text {a }}$ The misalignment tolerance is related to the length of arm.
    ${ }^{\mathrm{b}}$ The misalignment tolerance is related to $R_{\mathrm{bfs}}$.

